THIS REPORT HAS BEEN DELIMITED AND CLEARED FOR PUBLIC RELEASE UNDER DOD DIRECTIVE 5200.20 AND NO RESTRICTIONS ARE IMPOSED UPON ITS USE AND DISCLOSURE.

DISTRIBUTION STATEMENT A

APPROVED FOR PUBLIC RELEASE; DISTRIBUTION UNLIMITED.

# AID 549 Armed Services Technical Information Agency

Reproduced by DOCUMENT SERVICE CENTER KNOTT BUILDING, DAYTON, 2, 0 HIO

This document is the property of the United States Government. It is furnished for the duration of the contract and shall be returned when no longer required, or upon recall by ASTIA to the following address: Armed Services Technical Information Agency, Document Service Center, Knott Building, Dayton 2, Ohio.

NOTICE: WHEN GOVERNMENT OR OTHER DRAWINGS, SPECIFICATIONS OR OTHER DATA ARE USED FOR ANY PURPOSE OTHER THAN IN CONNECTION WITH A DEFINITELY RELATED GOVERNMENT PROCUREMENT OPERATION, THE U. S. GOVERNMENT THEREBY INCURS NO RESPONSIBILITY, NOR ANY OBLIGATION WHATSOEVER; AND THE FACT THAT THE GOVERNMENT MAY HAVE FORMULATED, FURNISHED, OR IN ANY WAY SUPPLIED THE SAID DRAWINGS, SPECIFICATIONS, OR OTHER DATA IS NOT TO BE REGARDED BY IMPLICATION OR OTHERWISE AS IN ANY MANNER LICENSING THE HOLDER OR ANY OTHER PERSON OR CORPORATION, OR CONVEYING ANY RIGHTS OR PERMISSION TO MANUFACTURE, USE OR SELL ANY PATENTED INVENTION THAT MAY IN ANY WAY BE RELATED THERETO.

# UNCLASSIFIED

AD NO CLOS X STAN FILE COPY

### Columbia University in the City of New York

DEPARTMENT OF CIVIL ENGINEERING AND ENGINEERING MECHANICS

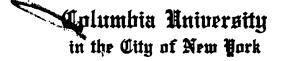
ERING 499



AN EXTENSION OF THE THEORY OF WATER HAMMER

by Richard S

AFSWP No. 922
Office of Naval Research
Contract No. Nonr-266(08)
Technical Report No. 15
CU-15-55-ONR-266(08)-CE
March 1955



DEPARTMENT OF CIVIL ENGINEERING AND ENGINEERING MECHANICS



# AN EXTENSION OF THE THEORY OF WATER HAMMER

by

Richard Skalak

AFSWP No. 922
Office of Naval Research
Contract No. Nonr-266(08)
Technical Report No. 15
CU-15-55-ONR-266(08)-CE
March 1955



## The American Society of Mechanical Engineers

29 WEST 39TH STREET, NEW YORK 18, NEW YORK

AN EXTENSION OF THE THEORY OF WATER HAMMER

R. Skalak
Asst. Prof. of Civil Engrg.
Dept. of Civil Engrg. & Engrg. Mechanics
Columbia University
School of Engineering
New York 27, N. Y.

Contributed by the Hydraulic Division for presentation at the ASME Diamond Jubilee Spring Meeting, Baltimore, Md. - April 18-21, 1955. (Manuscript received at ASME Headquarters May 19, 1954.)

Written discussion on this paper will be accepted up to May 20, 1955.

(Copies will be available until Feb. 3, 1956)

ADVANCE COPY: Released for general publication upon presentation.

Decision on publication of this paper in our ASME journal had not been taken when this pamphlet was prepared. Discussion is printed only if the paper is published in an ASME journal.

Printed in U.S.A.

Price: 50 cents per copy

(25 cents to ASME members)

### AN EXTENSION OF THE THEORY OF WATER HAPPIER

### By R. Skalak

### NOMENCLATURE

The following nomenclature is used in the paper:

c	Velocity of sound in fluid
c <sub>e</sub>	Joukowsky water hammer velocity (See eq. 1)
<b>K</b> .	Bulk modulus of fluid
a	Radius of tube
h	Thickness of tube wall
E	Young's modulus of tube wall material
p	Pressure in fluid
<b>A</b> '	Axial velocity of fluid
z	Distance axially along the tube
t	Time
u	Axial displacement of tube
W	Radial deflection of tube wall (+ outward)
Po	Initial density of fluid
r	Radial distance
ф	Velocity potential
m	Mass of tube per unit surface area
7	Poisson's ratio for tube wall
co	$\sqrt{\frac{Eh}{m(1-V^{-2})}}$ Velocity of sound in tube wall
q	Force per unit area on tube wall
٤, ۍ	Transform variables
n,s	Integers
N	See eq. (23)
$\triangle$	See eq. (24)
J <sub>0</sub> ,J <sub>1</sub>	Bessel functions
i =	√ <del>-1</del>
D	See eq. (30)
F	See eq. (36)
G	See eq. (37)
_	~ (00)

See eq. (38)

В

c',c Arbitrary constant velocities I, H Integrals. See eq. (53)

c<sub>1</sub>,c<sub>2</sub> Phase velocities

L Length of wave front

c<sup>2</sup>

z' See eq. (48)

R = 
$$\frac{c^2}{2}$$

Variable of integration

 $\beta$  See eq. (50)

A =  $\frac{\rho_0 a}{m}$ 

### 1. Introduction

The usual theory of water hammer which is used to compute the magnitude and velocity of pressure waves in a pipe filled with an elastic fluid was given by Joukowsky [8] in 1898. This theory predicts that pressure waves travel without change of shape at a velocity c:

$$C_e = \frac{C}{\sqrt{1 + \frac{2Ka}{Eh}}} \tag{1}$$

In the derivation of this formula it is assumed that the pressure is uniform across any section of the pipe and that the deflection of the pipe is equal to the static deflection due to the instantaneous pressure in the fluid. The first assumption neglects, in effect, the inertial forces associated with radial motion of the fluid. The second neglects the mass of the pipe wall and the longitudinal and bending stresses in the pipe wall.

For the case of an infinite train of sinusoidal pressure waves in a tube filled with fluid, much work has been done without resorting to the above assumptions. In 1878 Korteweg [9] showed that the phase velocity of sinusoidal waves varies with the wave-length when the radial inertia of the fluid and the mass of the pipe wall are taken into account. Korteweg also gave the expression for c<sub>e</sub>, eq. (1) as an approximation of the phase velocity for long wave-lengths. The work of Korteweg was extended in 1898 by Lamb [10] who considered the pipe as an elastic membrane thus including the effect of longitudinal stresses in the pipe wall. Lamb showed there are two finite phase velocities as the wave length approaches infinity.

One of these is very close to Korteweg's result c<sub>e</sub> and the other is very nearly the velocity of sound in the material of the tube wall. Recent papers by Jacobi [6] and Thomson [14] include the flexural stiffness of the pipe wall. The inclusion of the bending stresses affects the phase velocities appreciably only for very short wave-lengths.

wave-lengths. In which was a state of the paper.

In the present paper water hammer waves are considered without some of the simplifications made in the Joukowsky theory. The equations of motion for a thin cylindrical tube are used, cf. Flügge [11],

with one additional term for the effect of rotatory inertia. The fluid is assumed to be elastic and inviscid but no assumptions are made regarding its motion except that it is small.

Only the case where the velocity of sound in the material of the pipe wall is greater than in the fluid will be considered. Much of the analysis would be valid in the opposite case, but the relative speeds of the various waves would differ.

### 2. A Formal Solution by Integral Transforms

obtained in the form of a double integral.

In this section a formal solution of a typical water hammer problem is The solution is reduced to terms of single real integrals whose evaluation is discussed in the next section.

The problem considered is to find the motion of a thin cylindrical tube filled with fluid starting from given initial conditions. The conditions, shown in Fig. 1, are that for t = 0:

$$p = 0, v = 0 \text{ for } z > 0$$
  
 $p = p_0, v = v_0 \text{ for } z < 0$   
 $u = 0, w = 0, \text{ for all } z$ 
(2)

The velocity v is assumed to be related to the pressure p by:

$$V_o = \frac{p_o}{\rho_o c} \tag{3}$$

This is the velocity that would exist behind a plane shock wave of pressure  $p_0$  in an infinite body of fluid. Thus the initial conditions for the fluid correspond to a step shock wave moving in the positive direction. An external pressure equal to  $p_0$  is assumed acting on the outside of the tube for z < 0 to prevent any motion of the tube at  $z = -\infty$ . This external pressure is assumed to remain on the tube for z < 0 for all t.

These initial conditions are intended to simulate the situation arising when a pipe leading from a reservoir is filled with fluid at rest below reservoir pressure and a valve at the reservoir end is suddenly opened. The part of the tube

left of the origin in Fig. 1 is a convenient device for supplying fluid and energy in place of a reservoir. It is expected that the particular means of generating a water hammer wave will not essentially affect its characteristics when the wave has progressed far enough down the pipe.

The motion of the fluid will be described by a velocity potential which must satisfy the equation:

$$\nabla^2 \Phi = \Phi_{rr} + \frac{1}{r} \Phi_r + \Phi_{z\bar{z}} = \frac{1}{c^2} \Phi$$
 (4)

where the subscripts r and z denote derivatives.

The motion of the tube due to a radial force per unit area, q(z, t), is governed by two equations:

$$\frac{\ddot{w}}{C_o^2} - \frac{h^2}{12C_o^2} \ddot{w}_{\frac{7}{2}\frac{7}{2}} + \left(1 + \frac{h^2}{12a^2}\right) \frac{w}{a^2} + \frac{v}{a} u_{\frac{7}{2}} + \frac{11^2}{12} w_{\frac{7}{2}\frac{7}{2}\frac{7}{2}} - \frac{h^2}{12a} u_{\frac{7}{2}\frac{7}{2}} = \frac{q_F}{mc_o^2}$$
 (5)

$$\frac{\ddot{u}}{c_0^2} - u_{zz} - \frac{7}{a} W_z + \frac{h^2}{12a} W_{zzz} = 0$$
 (6)

detailed

A derivation of these equations is given in [12]. They agree with the equations given by Flügge [11] except that the rotatory inertia term  $\ddot{w}_{zz}$  has been added above.

The fact that the fluid and tube are expected to remain in contact is expressed by:

$$\dot{w} = -\left[\frac{\partial \phi}{\partial r}\right]_{r=a} \tag{7}$$

The pressure exerted on the tube wall by the fluid is:

$$\left[\rho\right]_{r=a} = \rho_o \left[\frac{\partial \phi}{\partial t}\right]_{r=a} \tag{8}$$

This is the value of q to be used for z>0 in eq. (5). For z<0, the external pressure -  $p_0$  must be added.

It is convenient to consider the solution of the problem as the sum of two parts:

The first part is the shock wave that would be produced by the initial conditions if the tube were rigid. The tube displacements  $w_1$ ,  $u_1$ , for this part are zero. The velocity and pressure  $v_1$ ,  $p_1$ , for the fluid are:

$$\begin{cases} V_1 = V_0 \\ P_1 = P_0 \end{cases} \text{ for } Z < ct$$

$$\begin{cases} V_1 = 0 \\ P_1 = 0 \end{cases} \text{ for } Z > ct$$

$$(9)$$

The second part is the response of the tube and fluid to a progressive load pressure  $p_s$  equal to the pressure produced by the shock wave of the first part. This load pressure acts on the tube wall radially outward. Its magnitude is  $p_o$  for 0 < z < ct and zero elsewhere.

Inasmuch as  $w_1$ ,  $w_1$ ,  $v_1$ , etc. are known, the problem reduces to finding the the response  $w_2$ ,  $w_2$ ,  $v_2$ , etc. of the tube and fluid to the load pressure  $p_s$ . For this second part of the solution the system is initially at rest and the governing equations are (4), (5), (6), (7) with:

$$q_{r_2} = \rho_0 \left[ \frac{\partial \phi}{\partial t} \right]_{r=a} + \rho_s \tag{10}$$

The solution is begun by transforming the governing equations and eliminating the velocity potential  $\phi_2$ . The remaining transformed equations are solved simultaneously and the inversions of the transformations then yield the solution.

With respect to z a Fourier transform, cf. [15] p. 42, is used. Defining the transform of a function g (z, t) with respect to z by:

$$\overline{g}(\xi,t) = \frac{1}{2\pi} \int_{-\infty}^{\infty} g(\xi,t) e^{-i \xi \xi} d\xi$$
 (11)

then the inversion is:

$$g(z,t) = \int_{-\infty}^{\infty} \overline{g}(\xi,t) e^{i\xi^{z}} d\xi$$
 (12)

With respect to time a Laplace transform is used, but the transform variable adopted is  $-i = \sqrt{-1}$  times the usual transform variable. This results in a certain symmetry with the first transform and convenience in later analysis. Defining the

transform of a function  $\ddot{g}$  ( $\xi$ , t) with respect to t by:

$$\overline{\overline{g}}(\xi, \Omega) = \frac{1}{2\pi} \int_{0}^{\infty} \overline{g}(\xi, t) e^{-i\Omega t} dt$$
 (13)

then the inversion is:

$$\overline{q}(\xi,t) = \int_{-\infty-i\alpha}^{\infty-i\alpha} \overline{q}(\xi,\mathbf{n}) e^{i\Omega t} d\Omega \qquad (14)$$

The double transform of derivatives of g(z, t) may be expressed in terms of the double transform of g(z, t) by the usual partial integration procedure, [13] p. 27. Using the fact that initial velocities and displacements are zero, the transforms of the derivatives involved are:

$$\frac{\overline{\partial^n}}{\partial z^n} \frac{\partial^s}{\partial t^s} \left[ q(z,t) \right] = (i\xi)^n (i\Omega)^s \, \overline{\overline{q}}(\xi,\Omega) \tag{15}$$

Using  $q_2$  as given by eq. (10), the double transformation of eqs. (4), (5), (6) and (7) yields:

$$\overline{\overline{\phi}}_{2rr} + \frac{1}{r}\overline{\overline{\phi}}_{2r} - \xi^2\overline{\overline{\phi}}_2 = -\frac{\Omega^2}{c^2}\overline{\overline{\phi}}_2$$
 (16)

$$-\frac{\Omega^{2}}{C_{0}^{2}}\overline{w}_{2} - \xi^{2}\Omega^{2}\frac{h^{2}}{12C_{0}^{2}}\overline{w}_{2} + \left(1 + \frac{h^{2}}{12a^{2}}\right)\frac{\overline{w}_{2}}{a^{2}} + i\xi\frac{\nu}{a}\overline{u}_{2} + \frac{\xi^{2}h^{4}}{12}\overline{w}_{2} + i\xi^{3}\frac{h^{2}}{12a}\overline{u}_{2} =$$

$$= i\Omega\frac{\rho_{0}}{mc_{0}^{2}}\left[\overline{\varphi}_{2}\right]_{r=a} + \frac{\overline{p}_{S}}{mc_{0}^{2}}$$
(17)

$$-\frac{\Omega^2}{C_0^2}\overline{u}_2 + \xi^2\overline{u}_2 - i\xi\frac{\nu}{a}\overline{w}_2 - i\xi^3\frac{h^2}{12a}\overline{w}_2 = 0$$
 (18)

$$i\Omega \overline{\overline{W}}_{2} = -\left[\overline{\overline{\Phi}}_{2r}\right]_{r=a} \tag{19}$$

The double transforms are functions of  $\xi$  and  $\Omega$  only except for  $\overline{\varphi}_2$  which is  $\overline{\varphi}_2$  ( $\xi$ ,  $\Omega$ , r). From eq. (16) it is found that the r dependence of  $\overline{\varphi}_2$  involves a Bessel function of the first kind. The function  $p_s$  is known. Solving eqs. (16), (17), (18), (19 for  $\overline{w}_2$ ,  $\overline{u}_2$ ,  $\overline{\varphi}_2$  and inverting yields:

$$W_{2} = -\frac{p_{o}}{4\pi^{2}m} \int_{-\infty}^{\infty} e^{i\frac{\pi}{2}} d\xi \int_{-\infty-i\alpha}^{\infty-i\alpha} \frac{e^{i\Omega t}}{\Omega^{3} \left(\frac{\Omega}{c} + \xi\right) \left[\frac{\rho_{o}\alpha}{2m} \Delta + \frac{1}{N}\right]} d\Omega$$
 (20)

$$u_{2} = \frac{p_{o}}{4\pi^{2}m} \int_{-\infty}^{\infty} e^{i\xi z} d\xi \int_{-\infty-i\alpha}^{\infty-i\alpha} \frac{\nu i \xi \left(1 + \frac{h^{2}}{12\nu} \xi^{2}\right) e^{i\Omega t}}{a \left(\frac{\Omega^{2}}{C_{o}^{2}} - \xi^{2}\right) \Omega^{3} \left(\frac{\Omega}{C} + \xi\right) \left[\frac{\rho_{o} a}{2m} \Delta + \frac{1}{N}\right]} d\Omega$$
 (21)

$$\Phi_{2} = -\frac{p_{o}}{4\pi^{2}m} \int_{-\infty}^{\infty} e^{i\xi^{2}} d\xi \int_{-\infty-i\alpha}^{\infty-i\alpha} \frac{i J_{o}(r\sqrt{\frac{\Omega^{2}}{c^{2}} - \xi^{2}}) e^{i\Omega t} d\Omega}{\int_{-\infty-i\alpha}^{2} \frac{\Omega^{2}}{C^{2}} - \xi^{2}} J_{i}(a\sqrt{\frac{\Omega^{2}}{c^{2}} - \xi^{2}}) \left(\frac{\Omega}{C} + \xi\right) \left[\frac{\rho_{o}a}{2m}\Delta + \frac{1}{N}\right]}$$

where:  

$$\frac{1}{N} = -1 + \frac{\Omega^{2} - c_{o}^{2}(1 - V^{2}) \xi^{2}}{\Omega^{2} a^{2}(\frac{\Omega^{2}}{C_{o}^{2}} - \xi^{2})} + \frac{h^{2}}{12} \left[ -\xi^{2} + \frac{c_{o}^{2}}{a^{4}\Omega^{2}} + \frac{2 c_{o}^{2} V \xi^{4}}{\Omega^{2} a^{2}(\frac{\Omega^{2}}{C_{o}^{2}} - \xi^{2})} + \frac{c_{o}^{2} \xi^{4}}{\Omega^{2}} + \frac{c_{o}^{2} k^{4}}{\Omega^{2}} + \frac{c_{o}^{2} h^{2} \xi^{6}}{12 \Omega^{2} a^{2}(\frac{\Omega^{2}}{C_{o}^{2}} - \xi^{2})} \right]$$

$$\Delta = \frac{2 J_{o} \left( a \sqrt{\frac{\Omega^{2}}{c^{2}} - \xi^{2}} \right)}{a \sqrt{\frac{\Omega^{2}}{2} - \xi^{2}}} J_{1} \left( a \sqrt{\frac{\Omega^{2}}{2} - \xi^{2}} \right) \tag{24}$$

The evaluation of the radial deflection  $w_2$  will be considered first. It is typical of the quantities of interest. The inner or  $\Omega$  integral of eq. (20) may be evaluated by Cauchy's residue theorem applied in the complex  $\Omega$  plane. During this process  $\xi$  is regarded as a constant real parameter since the outer integral involves only real values of  $\xi$ .

The factor  $\Omega^3$  in the denominator of the integrand in eq. (20) produces a simple pole at  $\Omega=0$  because  $(\Omega^2\frac{1}{N})$  is finite at  $\Omega=0$ . There is no pole at  $(\Omega+\xi)=0$  because  $(\Omega+\xi)\Delta$  is finite at this point. There are poles at the points for which the remaining bracket is zero, i.e., where:

$$\left[\frac{\rho_0 \alpha}{2 \, \text{m}} \Delta + \frac{1}{N}\right] = 0 \tag{25}$$

For any real  $\xi$  this equation has an infinite number of roots  $\Omega_n$ . These roots have an interesting physical significance. They are the circular frequencies of the modes of free vibration of the tube and fluid system having a wave-length  $\lambda = \frac{2\pi}{\xi}$ . Alternatively the ratios  $\frac{\Omega_n}{\xi}$  are the phase velocities  $c_n$  of progressive sinusoidal waves of wave-length  $\frac{2\pi}{\xi}$ . These facts may be developed by considering the governing equations (4), (5), (6) and (7) with q in eq. (5) equal to the fluid pressure only. A solution is assumed of the form:

$$W = w' e^{i(\xi z + \Omega t)}$$

$$u = u' e^{i(\xi z + \Omega t)}$$

$$\phi = \phi' f(r) e^{i(\xi z + \Omega t)}$$
(26)

where the prime quantities are constants and f(r) is an unknown function of r. Substituting eqs. (26) in eqs. (4), (5), (6) and (7) the constants and f(r) may be eliminated, leading to exactly the transcendental eq. (25) relating  $\{$  and  $\Omega$ .

For real values of  $\S$  all the roots  $\Omega_n$  of eq. (25) are also real. If any of the  $\Omega_n$  were complex or imaginary the corresponding free vibration would grow or decay indefinitely in time. Since the system has no means of dissipating or radiating energy, this is not possible physically.

Fig. 2 shows the  $\Omega_n$  vs  $\S$  values for a typical case in which the velocity of sound in the tube wall is greater than in the fluid, i.e., c > c. It is found that for any value of  $\S$  there is one root  $\Omega_1$ , for which  $(\frac{\Omega^2}{c^2} - \S^2)$  is negative. This root corresponds to a phase velocity less than c, the velocity of sound in the fluid. There are also an infinite number of roots for which  $(\frac{\Omega^2}{c^2} - \S^2)$  is positive. These correspond to phase velocities greater than c. Fig. 3 shows the same data as in Fig. 2 plotted as phase velocity vs wave-length.

The fact that two sets of  $\Omega_n$  curves pass through the origin in Fig. 2 shows that there are two modes which have finite phase velocities as the wave length approaches infinity and the frequency approaches zero. This was also shown by Lamb [10], p. 7. It is mentioned here because in a recent paper [14], p. 933, it is erroneously stated that there is a cutoff frequency below which only one mode may be propagated.

The above discussion shows that the poles of the integrand of eq. (20) are at  $\Omega$  = 0 and  $\Omega$  =  $\Omega_{-n}$ , all on the r.al  $\Omega$  axis. The path of integration is below this axis and it will be closed in the usual manner by adding a semicircle of infinite radius in the upper half of the  $\Omega$  plane. The integral over the semicircle approaches zero as the radius approaches infinity. The contribution of the pole at  $\Omega = 0$  to  $w_2$  is:

$$W_{2a} = \frac{p_o}{2\pi i m} \begin{cases} \frac{e^{i\xi^2}}{\left(1 - V^2 + \frac{h^2}{12a^2}\right) - \frac{c_o^2 V h^2}{6a^2} \xi^2 + \frac{h^2 c_o^2}{12} \left(1 - \frac{h^2}{12a^2}\right) \xi^4} \end{cases}$$
 (27)

The contributions of the  $\Omega_n$  poles to  $w_2$  may be expressed as:

$$w_{2b} = -\frac{p_o}{4\pi^2 m} \int_{-\infty}^{\infty} e^{i\xi^{\frac{7}{2}}} 2\pi i \sum_{\Omega_n} \left[ \frac{e^{i\Omega t}}{\Omega^3 \left(\frac{\Omega}{c} + \xi\right) \frac{d}{d\Omega} \left[\frac{\rho_o a}{2m} \Delta + \frac{1}{N}\right]} \right] d\xi \qquad (28)$$

The sum indicated is over the  $\Omega_n$  roots for each value of  $\xi$  . The values of the Ω as functions of \( \frac{1}{2} \) are given by Fig. 2. Eq. (28) may be written in the form

$$W_{2b} = \frac{-p_o}{2\pi i m} \int_{-\infty}^{\infty} \frac{e^{i(\xi z + \Omega t)}}{\left(\frac{\Omega}{c} + \xi\right)D} d\xi$$
 (29)

where 
$$D = -\Omega^{3} \frac{d}{d\Omega} \left[ \frac{\rho_{o} a}{2 m} \Delta + \frac{1}{N} \right] = \Omega^{3} \left\{ \frac{\rho_{o} \Omega a}{m c^{2} \left(\frac{\Omega^{2}}{c^{2}} - \xi^{2}\right)} \left[ 1 + \frac{J_{o}^{2}}{J_{i}^{2}} \right] + \frac{2 C_{o}^{2}}{\Omega^{3} a^{2}} \left[ \frac{\nu^{2}}{\left(1 - C_{o}^{2} - \frac{\xi^{2}}{\Omega^{2}}\right)^{2}} + 1 - \nu^{2} + \frac{h^{2} \left(1 + a^{4} \xi^{4}\right)}{12 a^{2}} + \frac{h^{2} \xi^{4} \left(2 \frac{\Omega^{2}}{C_{o}^{2}} - \xi^{2}\right)}{12 \left(\frac{\Omega^{2}}{C_{o}^{2}} - \xi^{2}\right)^{2}} \left(2 \nu + \frac{h^{2} \xi^{2}}{12}\right) \right]$$

The integral in eq. (29) covers all four quadrants of the  $\hat{\Lambda}_n$  curves in Fig. 2, but it may be expressed as an integral over the first quadrant values only because D is even in  $\Omega$  and even in  $\xi$  and the  $\Omega$  curves are symmetric about both axes. Thus:

$$W_{2h} = \frac{p_o}{\pi m} \int_{0}^{\infty} \sum_{\Omega_n} \left[ \frac{\sin(\xi z - \Omega t)}{\left(\frac{\Omega}{C} - \xi\right)D} - \frac{\sin(\xi z + \Omega t)}{\left(\frac{\Omega}{C} + \xi\right)D} \right] d\xi$$
 (31)

where the summation is over the  $\Omega_{-}$  curves in only the first quadrant of Fig. 2. This concludes the fermal solution for w. The evaluation of the integrals remaining in eqs. (27) and (31) will be considered in the next section.

instead of evaluating  $u_2$  and  $\phi_2$ , certain quantities of interest which are

derivatives of  $u_2$  and  $\phi_2$  will be computed directly. These are:

Pressure in the fluid: 
$$p_2 = Q \frac{\partial \phi_2}{\partial t}$$
Axial velocity of the fluid: 
$$v_2 = -\frac{\partial \phi_2}{\partial t}$$
Longitudinal strain in the tube: 
$$u_{\pm 2} = \frac{\partial u_2}{\partial \pm}$$

Expressions for  $p_2$ ,  $v_2$  and  $u_{z2}$  are derived from the double integrals for  $u_2$  and  $\phi_2$ , eqs. (21) and (22), through application of eq. (15). The inner integral in  $\Omega$  may be evaluated in each case by using Cauchy's residue theorem as in the case of  $w_2$ . The results are:

$$p_{2} = \frac{p_{o}}{2\pi i} \int_{-\infty}^{\infty} \frac{e^{i\xi(z-ct)}}{\xi} d\xi + \frac{p_{o}\rho_{o}}{\pi m} \int_{0}^{\infty} \frac{\sin(\xi z - \Omega t)}{\left(\frac{\Omega}{C} - \xi\right)F} - \frac{\sin(\xi z + \Omega t)}{\left(\frac{\Omega}{C} + \xi\right)F} d\xi$$
 (33)

$$V_{2} = \frac{P_{o}}{2\pi i P_{o}C} \int_{-\infty}^{\infty} \frac{e^{i\xi(z-ct)}}{\xi} d\xi + \frac{P_{o}}{\pi m} \int_{0}^{\infty} \frac{\sin(\xi z - \Omega t)}{\left(\frac{\Omega}{C} - \xi\right)G} + \frac{\sin(\xi z + \Omega t)}{\left(\frac{\Omega}{C} + \xi\right)G} d\xi$$
 (34)

$$u_{\frac{2}{2}} = \frac{i p_o V}{2 \pi m a} \int_{-\infty}^{\infty} \frac{\left(1 + \frac{h^2 \xi^2}{12 V}\right) e^{i \xi^2}}{f(\xi)} d\xi + \frac{p_o V}{\pi m a} \int_{0}^{\infty} \frac{\left[\frac{\sin(\xi z - \Omega t)}{C} - \frac{\sin(\xi z + \Omega t)}{C}\right] d\xi}{\left(\frac{\Omega}{C} - \xi\right) B} - \frac{\sin(\xi z + \Omega t)}{\left(\frac{\Omega}{C} + \xi\right) B} d\xi$$
 (35)

where:

$$F = -\frac{D}{\Omega^{2}} \sqrt{\frac{\Omega^{2} - \xi^{2}}{C^{2} - \xi^{2}}} \frac{J_{i} \left( a \sqrt{\frac{\Omega^{2} - \xi^{2}}{C^{2} - \xi^{2}}} \right)}{J_{o} \left( r \sqrt{\frac{\Omega^{2} - \xi^{2}}{C^{2} - \xi^{2}}} \right)}$$
(36)

$$G = F \frac{\Omega}{\xi}$$
 (37)

$$B = + \frac{D\left(\frac{\Omega^2}{C_o^2} - \xi^2\right)}{\xi^2 \left(1 + \frac{h^2}{12V}\xi^2\right)}$$
(38)

and  $f(\xi)$  stands for the denominator of the integrand in eq. (27).

The solution of the problem originally posed at the beginning of this section is given formally by:

$$W = W_{2a} + W_{2b} \qquad Eqs. (27) \text{ and } (31)$$

$$p = p_{1} + p_{2} \qquad Eqs. (9) \text{ and } (33)$$

$$V = V_{1} + V_{2} \qquad Eqs. (9) \text{ and } (3l_{1})$$

$$u_{2} = u_{22} \qquad Eq. (35)$$
(39)

### 3. Evaluation of the Solution for Large Values of |z| and t.

In the formal solution indicated by eqs. (39) the integrations with respect to  $\xi$  would be difficult to carry out in general because the  $\Omega_n$  values are known as functions of  $\xi$  only through the transcendental equation (25). However, some information may be obtained by considering approximations of the integrals for large values of |z| and t.

Consider the part of w given by eq. (31) and, in particular, values of s and t such that:

$$z = c' t (40)$$

where c' is an arbitrarily selected velocity. Substituting in eq. (31):

$$W_{2b} = \frac{p_o}{\pi m} \int_{\Omega_n}^{\infty} \left[ \frac{\sin(\xi c' - \Omega)t}{\left(\frac{\Omega}{c} - \xi\right)D} - \frac{\sin(\xi c' + \Omega)t}{\left(\frac{\Omega}{c} + \xi\right)D} \right] d\xi$$
 (41)

It is found by inspection of D, that the functions  $\frac{1}{(\frac{\Omega}{c} - \frac{\xi}{c}) D}$  and  $\frac{1}{(\frac{\Omega}{c} + \frac{\xi}{c}) D}$ 

are bounded for all of the  $\Omega_n$  curves for all  $\{$  except for the lowest two curves,  $\Omega_1$  and  $\Omega_2$ , at the point  $\{$  = 0. The distinction between the lowest two and the remaining  $\Omega_n$  curves arises because only the  $\Omega_1$  and  $\Omega_2$  curves pass through the origin. A physical interpretation of this fact is that only these two lowest modes have finite phase velocities as the wave-length increases indefinitely.

For the higher  $\Omega_n$  branches, n > 2, the integral (41) may be approximated by the method of stationary phase [7] p. 505, for large values of t. This method shows

these integrals to be of the order of  $\frac{1}{\sqrt{1}}$ .

The integrals over the  $\Omega_1$  and  $\Omega_2$  curves will be broken into two parts; one part from  $\{=0 \text{ to } \{=\in \text{ and a second part from } \in \text{ to } \infty, \text{ where } \in \text{ is a small}$ positive number. By the stationary phase theorem the portions from 6 to  $\infty$  are again 0 ( $\frac{1}{\sqrt{t}}$ ). Collecting these results, eq. (41) becomes:

$$W_{2b} = \frac{p_o}{\pi m} \int_{\Omega_{1,2}}^{\xi} \frac{\left[\frac{\sin(\xi c' - \Omega)t}{\left(\frac{\Omega}{c} - \xi\right)D} - \frac{\sin(\xi c' + \Omega)t}{\left(\frac{\Omega}{c} + \xi\right)D}\right] d\xi + O\left(\frac{1}{\sqrt{t}}\right)$$
 (42)

Since ( may be chosen indefinitely small, the functions in the integrand may be approximated by an appropriate number of terms of their expansions about \{ = 0. Consider the first term of the integral with  $\Omega = \Omega_{1}$  which may be written:

$$W_{I} = \frac{p_{o}}{\pi m} \int_{0}^{\epsilon} \frac{\sin(\xi c' - \Omega_{i})t}{\xi(\frac{\Omega_{i}}{\xi c} - 1)D} d\xi$$
 (43)

The denominator of the integrand approaches a constant times { as { approaches zero. The constant is:

$$\lim_{\xi \to 0} \left( \frac{\Omega_1}{\xi^C} - 1 \right) D = \left( \frac{C_1}{C} - 1 \right) D_1 \tag{44}$$

where:  

$$C_{1} = \lim_{\xi \to 0} \left( \frac{\Omega_{1}}{\xi} \right)$$

$$D_{1} = \lim_{\xi \to 0} D(\Omega_{1}, \xi) = \frac{4 \cdot \rho_{0} \cdot C_{1}^{4}}{m \cdot c^{2} a \left( \frac{C_{1}^{2}}{C^{2}} - 1 \right)^{2}} + \frac{2 \cdot C_{0}^{2}}{a^{2}} \left[ \frac{\nu^{2}}{\left( 1 - \frac{C_{0}^{2}}{C_{1}^{2}} \right)^{2}} + 1 - \nu^{2} + \frac{h^{2}}{12a^{2}} \right]$$
(45)

The constant c, is the phase velocity of the slowest waves for wave-lengths approaching infinity. It is very close to the Joukowsky velocity  $\mathbf{c}_{\mathbf{a}}$ . An expression for c, is derived in the appendix.

In the numerator of eq. (43)  $\Omega_{\gamma}$  is replaced by the first two terms of its expansion about  $\xi = 0$ :

$$\Omega_1 = C_1 \xi - d_1 \xi^3 \tag{46}$$

This expansion is developed in the appendix. It is necessary to use two terms of the expansion because the first term alone would result in a vanishing numerator for the case  $c' = c_1$ .

Using eqs. (44) and (46) in (43) yields:

$$W_{I} = \frac{p_{o}}{\pi m \left(\frac{C_{i}}{C} - 1\right) D_{i}} \int_{0}^{\infty} \frac{\sin \left(\xi z' + d_{i} \xi^{3} t\right)}{\xi} d\xi$$
 (47)

where z' is defined by:

$$\Xi' = (c' - c_1) t \tag{48}$$

The new coordinate  $z^*$  is the location of any point relative to an origin moving at velocity  $c_1$ . In eq. (47) the upper limit has been changed from  $\epsilon$  to  $\infty$ . The integral thus added is  $O(\frac{1}{\sqrt{t}})$  by the stationary phase theorem, and is negligible in the present approximation.

For  $z^1 = 0$ , the integral in eq. (47) has a constant value of  $\frac{\pi}{6}$  provided d<sub>1</sub> is positive which is the case here. For points where  $|z^1|$  is large, the term containing  $\xi^3$  may be neglected. Then the integral in eq. (47) becomes  $\frac{1}{2}$  for  $\frac{1}{2}$ . For points near  $z^1 = 0$  it is convenient to change the variable of integration from  $\xi$  to  $\eta = \xi z^1$ . Then:

$$W_{I} = \frac{p_{o}}{\pi m \left(\frac{C_{I}}{C} - I\right) D_{I}} \int_{0}^{\infty} \frac{\sin \left(\eta + \beta \eta^{3}\right)}{\eta} d\eta \qquad (49)$$

where:

$$\beta = \frac{d_1 t}{z^{13}} \tag{50}$$

From eq. (49) it may be seen that for  $w_{\rm I}$  constant  $\beta$  must be constant, say  $\beta$  o. Then:

$$z' = \sqrt[3]{d_1 t \beta_0} \tag{51}$$

This shows that points for which  $\mathbf{w}_{\mathbf{I}}$  is constant continually move away from the point  $\mathbf{z}^{\dagger} = 0$ . To summarize,  $\mathbf{w}_{\mathbf{I}}$  is a wave having constant deflection at the point  $\mathbf{z}^{\dagger} = 0$  i.e., moving with velocity  $\mathbf{c}_{\mathbf{I}}$ . At points far ahead and far behind this wave front point the deflection is also constant, but the wave front continually spreads out

around the point z' = 0. Fig. 4 shows the general shape of the wave.

If the same reasoning that has been used to evaluate  $w_{\rm I}$  is applied to the remaining terms of eq. (42) similar results are obtained. Thus for large t:

$$W_{2b} = I_1 - H_1 + I_2 - H_2 \tag{52}$$

where:

$$\frac{I_{n}}{H_{n}} = \frac{p_{o}}{\pi m \left(\frac{C_{n}}{C} + 1\right) D_{n}} \int_{0}^{\infty} \frac{\sin(\Xi + C_{n}t \pm d_{n}\xi^{2}t)\xi}{\xi} d\xi \qquad n = 1, 2$$
 (53)

The constants  $c_2$ ,  $D_2$ ,  $d_2$  which appear in  $H_2$  and  $I_2$  are defined in the same way as  $c_1$ ,  $D_1$ ,  $d_1$  in eqs. (45) and (46) by changing all subscripts from 1 to 2 in these equations. The velocity  $c_2$  is very nearly the velocity of sound in the tube wall,  $c_0$ .

The term  $I_1$  is the deflection  $w_I$  discussed above. The  $H_1$  term is a wave of the same velocity and shape as  $w_I$  but of smaller applitude and traveling in the negative direction.

The  $H_2$  and  $I_2$  terms are also waves of the same shape as  $w_I$ , but traveling at the velocity  $c_2$ . These precursor waves are found to be very small compared to the main water hammer waves  $I_1$ ,  $H_1$  which travel at velocity  $c_1$ .

The complete deflection w is the sum of  $w_{2b}$ , eq. (52) and  $w_{2a}$ , eq. (27). The deflection  $w_{2a}$  is independent of time and its value for large values of |z| may be approximated by the same kind of reasoning as used for  $w_{2b}$  for large values of t. The result is:

$$W_{2a} = \pm \frac{p_0 a^2}{2Eh \left[1 + \frac{h^2}{12 a^2 (1 - \nu^2)}\right]} \quad \text{for } \pm 2$$
 (54)

The term  $\frac{h^2}{12a^2(1-v^2)}$  will be neglected compared to unity since only thinwalled tubes are considered. Thus:

$$W_{2a} = \pm \frac{p_0 a^2}{2Eh}$$
 for  $\pm z$  (55)

The procedures used to evaluate w may also be used to obtain approximations of p, v and  $u_z$  for large |z| and t starting with the general expressions, eqs. (39). The results are:

$$W = \pm \frac{p_{o}a^{2}}{2Eh} + I_{1} - H_{1} + I_{2} - H_{2} \qquad \text{for } \pm \pm \frac{p_{o}a^{2}}{2Eh} + I_{1} - H_{1} + I_{2} - H_{2} \qquad \text{for } \pm \pm \frac{p_{o}c_{2}^{2}}{a\left[\frac{C_{1}^{2}}{c^{2}} - I\right]} \left(I_{1} - H_{1}\right) - \frac{2\rho_{o}c_{2}^{2}}{a\left[\frac{C_{2}^{2}}{c^{2}} - I\right]} \left(I_{2} - H_{2}\right)$$

$$V = \frac{p_{o}}{2\rho_{o}c} - \frac{2c_{1}}{a\left[\frac{C_{1}^{2}}{c^{2}} - I\right]} \left(I_{1} + H_{1}\right) - \frac{2c_{2}}{a\left[\frac{C_{2}^{2}}{c^{2}} - I\right]} \left(I_{2} + H_{2}\right)$$

$$U_{2} = \pm \frac{p_{o}va}{2Eh} + \frac{v}{a\left[\frac{c_{1}^{2}}{C_{0}^{2}} - I\right]} \left(I_{1} - H_{1}\right) + \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{0}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I_{2} - H_{2}\right) \text{ for } \pm \frac{v}{a\left[\frac{C_{2}^{2}}{C_{2}^{2}} - I\right]} \left(I$$

These equations lead to four wave fronts in each case as discussed for w.

Numerical results for a typical case of water in a thin steel tube are shown in

Fig. 5. In the present approximation for large t, the pressure and fluid velocity

are again independent of r, i.e., they are constant across any section.

In order to estimate the length over which a given wave front extends, the slope w<sub>Z</sub> will be computed at the points which move with constant velocity c<sub>1</sub> or c<sub>2</sub>. The difference in w before and after a wave front divided by the slope at the wave front point will be indicative of the length of a wave front.

The double integral for  $w_z$  is derived from  $w_z$  eq. (20). The first integral may be evaluated and the result folded by the same procedures used for  $w_z$ . For a point moving so that  $z = c_1$  t:

$$W_{z} = \frac{p_{o}}{\pi m} \int_{0}^{\infty} \frac{\xi \cos \xi c_{i}t}{f(\xi)} d\xi + \frac{p_{o}}{\pi m} \int_{0}^{\infty} \left[ \frac{\xi \cos (c_{i}\xi - \Omega)t}{\left(\frac{\Omega}{C} - \xi\right)D} - \frac{\xi \cos (c_{i}\xi + \Omega)t}{\left(\frac{\Omega}{C} + \xi\right)D} \right] d\xi$$
 (57)

The function  $\frac{\xi}{f(\xi)}$  is bounded for all  $\xi$  and the first integral in eq. (57) is 0  $(\frac{1}{\xi})$  by the Riemann-Lebesgue lemmas [16] p. 172.

In the second integral the integrand is also bounded. This integral may be approximated by the method of stationary phase. The dominant part of the integral will come from the vicinity of stationary phase points which are defined by:

$$\frac{d}{d\xi}(\xi c_1 \pm \Omega) = 0 \qquad \text{i.e.} \qquad \frac{d\Omega}{d\xi} = \pm c_1 \qquad (58)$$

In general such points give a contribution which is 0  $(\frac{1}{\sqrt{t}})$ , but if:

$$\frac{d^2}{d\xi^2} \left( \xi C_1 \pm \Omega \right) = 0 \qquad \text{i.e.} \quad \frac{d^2 \Omega}{d\xi^2} = 0$$
 (59)

then the contribution to the integral from the vicinity of this point is 0  $(\frac{1}{\sqrt[3]{t}})$ , [5] p. 40. Eqs. (58) and (59) are both satisfied on the  $\Omega_1$  curve at  $\xi = 0$ . Hence this contribution will predominate over all others for large values of t.

To evaluate the dominant term, the expression for  $\Omega_1$ , eq. (46), is substituted in eq. (57). Hence for large t:

$$W_{z} = \frac{p_{o}}{\pi m} \int \frac{\cos(d_{1}\xi^{3}t)}{\left(\frac{C_{1}}{C}-I\right)D_{1}} d\xi$$
 (60)

Changing the variable of integration to  $\eta = \left\{ \left( \sqrt[3]{d_1 t} \right) : \right\}$ 

$$W_{\frac{3}{2}} = \frac{p_o}{\pi m \left(\frac{C_1}{C} - 1\right) D_1 \sqrt[3]{d_1 t}} \int_{0}^{\infty} \cos \eta^3 d\eta$$
 (61)

This integral (61) is known, [5] p. 41. Hence:

$$W_{\frac{2}{2}} = \frac{p_o \left(\frac{1}{3}\right) \sin \frac{\pi}{3}}{3\pi m \left(\frac{c_1}{C} - 1\right) D_i \sqrt[3]{d_i t}}$$
 (62)

Now the change of w from a location far behind the point  $z = c_1$  t to a location of the point far ahead,  $z = c_1$  t is by eq. (49):

$$\Delta W = \frac{p_o}{m\left(\frac{c_1}{C} - I\right)D_I} \tag{63}$$

Defining the "nominal length" L of the wave front to be  $\Delta w$  divided by  $w_z$ :

$$L_1 = \frac{3\pi^3 \sqrt{d_1 t}}{\Gamma(\frac{1}{3}) \sin \frac{\pi}{3}}$$
 (64)

Computation of the "nominal length",  $L_2$ , of the wave fronts moving at velocity  $c_2$  results in the same formula with  $d_1$  replaced by  $d_2$ .

Some numerical values of  $L_1$  and  $L_2$  are given in Table 1. It may be noted that although the length of the wave front is long compared to the diameter of the pipe it is short compared to the distance from the origin that the wave has traveled.

The above computations of the lengths of wave fronts hold also for p, v and  $u_x$  because they are described by the same integrals,  $I_1$ ,  $H_1$ ,  $I_2$ ,  $H_2$ .

### 4. An Extension of Joukowsky's Method

Some of the results derived by use of integral transforms may also be derived by a much simpler procedure which is an extension of Joukowsky's method. The extension consists of taking into account the longitudinal stresses and longitudinal inertia of the pipe wall. This makes possible the precursor type wave which is essentially a tension wave in the pipe wall.

The following derivation follows the general plan of Joukowsky's paper [8]. The pressure and axial velocity of the fluid are assumed to be uniform over any cross-section of the pipe. The fluid is assumed compressible, its bulk modulus being K. The equation of continuity may then be written in the form:

$$-\frac{\partial V}{\partial z} = \frac{1}{K} \frac{\partial P}{\partial t} + \frac{2}{a} \frac{\partial W}{\partial z}$$
 (65)

The equations of motion of the fluid reduce to:

$$\rho_0 \frac{\partial V}{\partial t} = -\frac{\partial P}{\partial z} \tag{66}$$

So far, these are the equations usually used. The new element is the use of the equations of motion of the tube considered as an elastic membrame instead of a ring of no mass. These equations are derived from eqs. (5) and (6) by dropping all terms arising from bending stiffness and rotatory inertia and substituting p for q:

$$\frac{w}{a^2} + \frac{v}{a} u_{\tilde{z}} = \frac{p}{mc_0^2} \tag{67}$$

$$\frac{\ddot{u}}{C_o^2} - u_{\frac{2}{2}} - \frac{\mathcal{V}}{a} w_{\frac{2}{2}} = 0 \tag{68}$$

In eq. (67) the term w has been omitted because the effect of radial inertia of the tube and of the fluid is neglected as in the Joukowsky theory.

No attempt will be made to solve the equations (65), (66), (67) and (68) in general. It is sufficient for the purpose at hand to show that the system permits solutions which are waves of arbitrary shape moving at either of the velocities c<sub>1</sub> or c<sub>2</sub> without dispersion. A solution is assumed of the form:

$$W = W' f(z + \overline{c}t)$$

$$P = P' f(z + \overline{c}t)$$

$$V = V' f(z + \overline{c}t)$$

$$U_{z} = U'_{z} f(z + \overline{c}t)$$

$$U_{t} = U'_{t} f(z + \overline{c}t)$$

$$(69)$$

where the primed quantities are constants and  $\bar{c}$  is an unknown constant velocity. The constants  $u_{\bar{t}}^1$ ,  $u_{\bar{z}}^1$  and  $\bar{c}$  are related by the requirement:

$$\frac{d}{dt} u_{\frac{2}{4}} = \frac{d}{d^{\frac{2}{4}}} u_{t} \tag{70}$$

Substituting the assumed form (69) into eqs. (65), (66), (67), (68) and (70), there result five linear homogeneous equations involving the five primed constants. In order that these equations have a solution different from zero it is necessary for the determinant of the coefficients to be zero. This condition yields an equation

involving  $\bar{c}$  which can be satisfied only if  $\bar{c}$  takes on the values  $\dot{c}_1$  or  $\dot{c}_2$ . Thus the velocities at which waves may travel without dispersion in the present consideration are the same as the principal velocities derived in the integral transform method.

A further point of interest is that the ratio of the change in w to the change in p across a wave front is the same in the extended Joukowsky theory and the transform solution. The same is true of any two of the quantities w, p, v,  $u_z$ , etc. This is shown by solving for the ratios of interest in each theory with  $c = c_1$  or  $c_2$  as the case may be.

These results are of interest because they indicate the significance of the assumptions made in the Joukowsky theory. It shows that c differs from c because longitudinal of the pipe. e the Joukowsky theory neglects longitudinal stresses and inertia. It also shows that the dispersion predicted by the transform method is due to the effects of the bending stiffness and of the radial inertia of the pipe and fluid. The effect of radial inertia is much more important. After the wave disperses somewhat, bending effects are negligible.

### 5. Other Initial Conditions

The integral transform method may be applied to certain other initial conditions beside those assumed in Sections 2 and 3. The problem and results shown in Fig. 6 illustrate a second example. The initial conditions are:

$$v = v_0$$
 for  $z < 0$   $v = -v_0$  for  $z > 0$   
 $p = u = w = u_t = w_t = 0$  for all  $z$ .

Under these conditions each half of the pipe simulates the water hammer problem in which an established flow is suddenly cut off by an instantaneous valve closure. Since by symmetry no fluid crosses the plane z = 0, each half of the pipe acts as a closed valve as far as the other half is concerned.

The magnitudes and velocities of the waves may also be established on the basis of the simplified procedure of Section 4. In general, this will be the simpler

method. It does not predict dispersion, but it may be assumed that the dispersion characteristics are the same in any case.

### 6. Conclusions

The theory developed herein shows the following results which are not contained in the usual theory:

- 1. If the initial conditions specify a sharp wave front of pressure, this front will gradually disperse over a finite length as it proceeds. This dispersion continues indefinitely so that no steady wave shape is ever reached.
- 2. Two waves will be generated in general; one which is similar to the Joukowsky result and one which is essentially a tension wave in the pipe wall. The pressure change due to this precursor is very small.
- 3. After sufficiently long time has elapsed since the generation of a wave there will be one point in each wave front which has a constant deflection and fluid pressure and moves at a constant velocity. This velocity is very close to the Joukowsky velocity for the main pressure wave. For the precursor this velocity is slightly less than the velocity of sound in the pipe wall.
- 4. The bending stresses in the pipe wall have very little effect on the characteristic velocities and the rates of dispersion after a sufficient time has elapsed. Such stresses can only be significant in the early stages of the motion in regions where the wave front is sharp so that the variation of pressure and deflection is very rapid along the length of the pipe.
- 5. The relations between fluid velocity, pressure and hoop stress given by the Joukowsky theory are substantially correct, but there are also longitudinal stresses in the pipe not predicted by the usual theory.

[2]

6. Available experimental data is in very close agreement with the predictions of the Joukowsky theory as to the pressure rise and velocity of the main pressure wave. Inasmuch as the analysis presented in this paper gives essentially the same results as the Joukowsky theory for these items, it is confirmed experimentally. However, there is no experimental data available on the precursor wave or the rate of dispersion predicted herein.

### ACKNOWLEDGEMENTS

The author is grateful to Professor H. H. Bleich for his advice and encouragement throughout the preparation of this paper and the thesis on which it is based. This research was partially supported by the Office of Naval Research under Contract Nonr 266(08).

### **LIBLIOGRAPHY**

- (1) Allievi, L., Allgemenie theorie uber die veranderliche Bewegung des Wassers in Leitungen, translated from Italian into German by R. Dubs and V. Bataillard, Julius Springer, Berlin, 1909.
- (2) American Society of Mechanical Engineers, Symposium on Water Hammer, 1933.
- (3) Biot, M. A., "Propagation of Elastic Waves in a Cylindrical Bore Containing Fluid," Journal of Applied Physics, Vol. 23, 1952, pp. 997-1055.
- (4) Fay R. D., "Waves in liquid-filled Cylinders," Journal of the Acoustical Society of America, Vol. 24, No. 5, Sept. 1952, pp. 459-462.
- (5) Havelock, T. H., The Propagation of Disturbances in Dispersive Media, Cambridge University Press, 1914.
- (6) Jacobi, W. J., "Propagation of Sound Waves along Liquid Cylinders," Journal of the Acoustical Society of America, Vol. 21, No. 2, 1949, pp. 120-127.
- (7) Jeffreys, H. and Jeffreys, B. S., Methods of Mathematical Physics, Cambridge University Press, 2nd Ed., 1950.
- (8) Joukowsky, N., "Uber den hydraulischen Stoss in Wasserleitungsrohren,"
  Memoires de l'Academie Imperiale des Sciences de St. Petersbourg, 8 serie,
  Vol. 9, No. 5, 1900.
- (9) Korteweg, D. J., "Weber die Fortpflanzungsgeschwindigkeit des Schalles in elastisches Rohren," Annalen der Physik und Chemie, 9 Floge, Band 5, 1878, pp. 525-512.

- (10) Lamb, H., "On the Velocity of Sound in a Tube, as Affected by the Elasticity of the Walls," Manchester Literary and Philosophical Society, Memoirs and Proc., Vol. 42, No. 9, 1898.
- (11) Flügge, W., Statik und Dynamik der Schalen, Julius Springer, Berlin, 1934, Reprinted by Edwards Brothers, Ann Arbor, Mich., 1943.
- (12) Skalak, R., An Extension of the Theory of Water Hammer, Doctoral thesis, Columbia University, 1954.
- (13) Sneddon, I. N., Fourier Transforms, McGraw-Kill, 1951.
- (14) Thomson, W. T., "Transmission of Pressure Waves in Liquid Filled Tuber," Proc. of the First U. S. National Congress of Applied Mechanics, ASME, 1951, pp. 927-933.
- (15) Titchmarsh, E. C., Introduction to the Theory of Fourier Integrals, Oxford University Press, 1937.
- (16) Whittaker, E. T. and Watson, G. N., A Course in Modern Analysis, Cambridge University Press, 4th Ed., 1952.

### APPENDIX

### Expansion of $\Omega_1$ and $\Omega_2$ About $\xi = 0$

The values  $\Omega_1$ ,  $\Omega_2$  are defined as the lowest two roots of eq. (25). From the numerical results represented by Fig. 2, it is expected that both  $\Omega_1$  and  $\Omega_2$  approach zero as  $\xi$  approaches zero. Hence,  $\Omega_1$  and  $\Omega_2$  have an expansion of the form:

$$\Omega = \bar{c} \xi - d \xi^3 + \dots \tag{71}$$

where only odd powers occur because the  $\Omega_n$  are odd functions of  $\xi$ . The values of  $\bar{c}$  and d are determined by substituting this expansion in eq. (25) and equating the coefficient of each power of  $\xi$  appearing to zero.

The terms  $\frac{1}{N}$  and  $\Delta$  in the transcendental equation (25) are given in full by eqs. (23) and (24). Expanding the Bessel functions,  $\Delta$  may be written in the form:

$$\Delta = \frac{\left[1 - \frac{\alpha^2}{4} \left(\frac{\Omega^2}{C^2} - \xi^2\right) + \frac{\alpha^4}{64} \left(\frac{\Omega^2}{C^2} - \xi^2\right)^2 + \cdots\right]}{\frac{\alpha^2}{4} \left(\frac{\Omega^2}{C^2} - \xi^2\right) \left[1 - \frac{\alpha^2}{8} \left(\frac{\Omega^2}{C^2} - \xi^2\right) + \frac{\alpha^4}{192} \left(\frac{\Omega^2}{C^2} - \xi^2\right)^2 + \cdots\right]}$$
(72)

In order to be able to clear eq. (25) of fractions, the quotient of the two infinite series in eq. (72) is written as a single series:

$$\Delta = \frac{1}{\frac{a^2}{4} \left( \frac{\Omega^2}{C^2} - \xi^2 \right)} \left[ 1 - \frac{a^2}{8} \left( \frac{\Omega^2}{C^2} - \xi^2 \right) - \frac{a^4}{192} \left( \frac{\Omega^2}{C^2} - \xi^2 \right)^2 + \cdots \right]$$
 (73)

Using this value the transcendental eq. (25) may be put in the form:

$$\frac{2\Omega^{2}}{ma} \left(\frac{\Omega^{2}}{c^{2}} - \xi^{2}\right) \left[1 - \frac{d^{2}}{8} \left(\frac{\Omega^{2}}{c^{2}} - \xi^{2}\right) - \frac{d^{4}}{192} \left(\frac{\Omega^{2}}{c^{2}} - \xi^{2}\right)^{2} + \cdots\right] + \\
+ \Omega^{2} \left(\frac{\Omega^{2}}{c^{2}} - \xi^{2}\right) \left(\frac{\Omega^{2}}{c^{2}} - \xi^{2}\right) \left[-1 - \frac{h^{2} \xi^{2}}{12}\right] + \\
+ \left(\frac{\Omega^{2}}{c^{2}} - \xi^{2}\right) \left(\frac{\Omega^{2}}{c^{2}} - \xi^{2}\right) \frac{h^{2} c_{o}^{2}}{12} \left[\frac{1}{a^{4}} + \xi^{4}\right] + \\
+ \left(\frac{\Omega^{2}}{c^{2}} - \xi^{2}\right) \frac{1}{a^{2}} \left[\Omega^{2} - c_{o}^{2} \left(1 - \nu^{2}\right) \xi^{2} + \frac{\nu h^{2} c_{o}^{2} \xi^{4}}{6} + \frac{c_{o}^{2} h^{4} \xi^{6}}{144}\right] = 0$$

Now the assumed expansion given by eq. (71) is substituted above for  $\Omega$ . The result is a series of the form:

$$b_1 \xi^4 + b_2 \xi^6 + b_3 \xi^8 + \cdots = 0$$
 (75)

Each coefficient  $b_n$  of this series must be zero since the value  $\Omega$  is a root of the transcendental eq. (25). Equating  $b_1$  to zero and neglecting  $\frac{h^2}{a^2}$  compared to unity yields an equation involving  $\bar{c}$  but not d. This is interpreted as an equation on the yet unknown  $\bar{c}$ . The roots are:

$$C_{1,1}C_{2} = C\left[\frac{2AR + R + R^{2}(1-V^{2}) + \sqrt{[2AR + R + R^{2}(1-V^{2})]^{2} - 4R^{2}(1-V^{2})(2A+R)}}{2(2A+R)}\right] (76)$$

where

$$R = \frac{C_0^2}{C^2} \qquad A = \frac{P_0 a}{m} \tag{77}$$

Equating  $b_2$  to zero yields an equation involving both  $\bar{c}$  and d. This equation is considered to determine two values of d, one for  $\bar{c} = c_1$  and one for  $\bar{c} = c_2$ . The solution of the equation b, = 0 for d yields:

$$d = ca^{2} \left[ \frac{(A+4)\left[\left(\frac{\bar{c}}{c}\right)^{5} - \left(\frac{\bar{c}}{c}\right)^{3}(1+R) + \left(\frac{\bar{c}}{c}\right)R\right] - \left(\frac{\bar{c}}{c}\right)\frac{h^{2}}{12a^{2}}\left(1+2R^{2}V\right)}{-16\left(\frac{\bar{c}}{c}\right)^{2}\left(2A+R\right) + 8R(2A+1) + 8R^{2}(1-V^{2})} \right]$$
(78)

(: ;		1	$\overline{}$	Π.		/									I									ı. J								·		$\mathbb{Z}$					
		1	1			/							\$	3	$\prod$			1	_			. (		Ţ.												Z			
				1	- :						Ŀ		tote	CLEV	$\prod$		1.							I			::								Z				$Z_{-}$
	V										Ī		2	Ü			• :	<i>i.</i>	. 4,	1						1-14		:		ار	$\angle$			1	1_		_		
	$\sum$		111				111		7	[			15gm	3		L			1		: :			<u>.</u>					1					<u> </u>					
	1	V:F				7	: :	::	1.	$\Delta$		Ŀ		Ľ							Fa			<u> </u>	Ŀ			1	1	: :	Ŀ	Ŀ	$\angle$	1			-/		
Ŀ		tI	<u> 4</u> 11				5		-	,	5		V_	Ŀ		1		L			4		1	_		::-	1	1						Ŀ	<u> </u>	_	L.		
L	L	5	+		1.	::	(ع)			L	(E)	1		L	1	=1	_	L	_	_			1	Ŀ	1	1		<u> </u>		.: .	<u> </u>	Ζ.			-	:	<u>/</u>		
	1		1	1::::				7			<u> </u>	7	_	Ľ		_	\_	_			3			Ľ,	1		L			L.:	1		1:	_		1	ر-	1 ···	
			٧	1				4	_		1:	-	7				1		1:4:	::	Va	1		2		1	11:			ļ.,	/_	:		_	سنبل	۲,			
		1	7	_	1		_	1	1		_				1	1	. \ \			111		<u>(+</u>	1	-::		<u>/</u>				/	_			ļ	_	4	-		$\dashv$
_	-	_	3						1		<u> </u>	_	:	1	1	122		$\downarrow$		1	1	2	1711	1	1	111		سببنا		/	-			1	ļ:_	1	-		
Ŀ	ļ			1::				:: 1	1		<u> </u>	١	-	A	<b>&gt;</b>	-	L.	13	γ.	/	2		-	ينيا	1		-				_		-	1	<u> </u>	<u> </u>	-		
-	1		1	Ŀ		_	1	Ŀ	4	-	1	<u> </u>	<del> </del>		<del> </del>	-		$\succeq$		K			1: 1	_	1	_	11;	1	١.,		ļ.,	•			1	L	╄-		
		do i																																					
<u> </u>	-	4	<del> </del>	Ĭ	•		-	9		•		Ţ.,,		٢,	+-	<b>!</b> -	-		1	1/2			-	2	f.	<u> </u>	<u>ا ::</u> ا	<u> </u>		۱.		·	-	<u> </u>	ļ	<u>-</u>	<del> </del>	F	
E			1		لننا	-		, 2			9.0	7/2		1			<u> </u>	4	1	7				-	1					1	-				-		-		
-		4	1		لننا			3			9.0	7/5					/	1	1	7			,	-	1					1			• • • • • • • • • • • • • • • • • • •				-	F	
			1		لننا			, , ,			93	1/2					/	1	1	7			\ \ \		1				,		-					7		F	
ļ. ļ.			1		لننا			, 30 ./			93	1/2					/	1	1	•			Z		1					ナナ <u> </u>	-				,	7	7	F	
- L			1								93	/						1		<b>f</b>			Z		7					+ + <u></u>	-						\ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \ \	F	
			1				-		1		9.9	/				7	/	1	1	•			\ \ \ \		7					ナナ <u> </u>	-								
			1						1			/				7		/		<b>f</b>					7					ナナ <u> </u>	-							F	
			/				1		1			/					/	/					ンフナナ							ナナ <u> </u>	-								
			1				/					/								\$- 1 <b>7</b> -										ナナ <u> </u>	-								
			/			7	1				/	/						7												ナナ <u> </u>	-								
			/			7					/	/						7												ナナ <u> </u>	-								
			/			7					/	/				/	/													ナナ <u> </u>	-								

Distance of wave front from origin, Feet

> Mominal length of wave front. Feet

> > Tites Seconds

Table I. Nominal Lengths of wave Fronts for Typical Example

3,218

19.1 32.6

4 2

Primary Mare

17,325 86,640

30.0 E.S.

Precursor Mave

Derived Values

# = 62.752 R = 32.478

61 - 0.64365

.2 - 3.4656 .6 - 3.532a

θ<sub>0</sub> = 1.94 slugs/rt<sup>3</sup> E = 30 x 10<sup>6</sup> psi F = 0.30

Fodulus of elasticity of steel:

Poisson's ratic for steel: Specific weight of steel! Ratio, A =  $\frac{P_0 a}{R}$ :

κ<sub>e</sub> - 450 lbe/rt<sup>3</sup>

c = 5,000 ft/mc

Velocity of sound in water:

Radius of pipe:

Density of water:

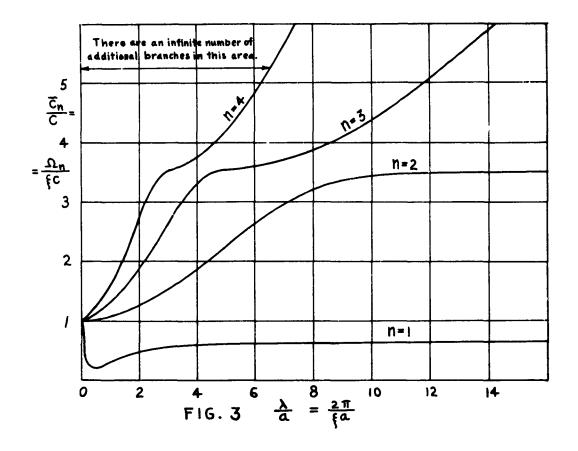
a = 1 foot

Assumed Values

62 = 0.64429

p=p, p=0 a
v=v, v=v
v=v
v=v
v=v
v=v
v=v

Fig.1



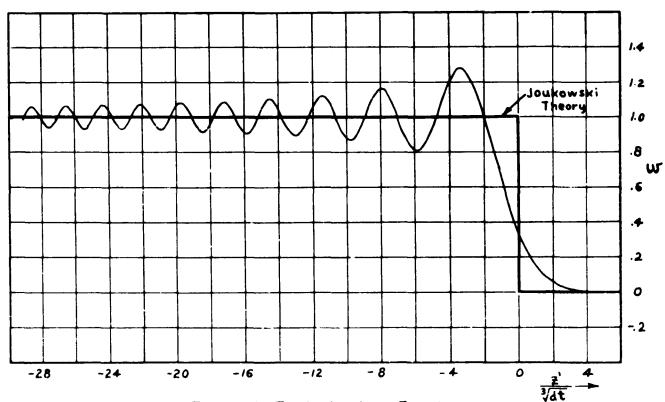


Fig. 4. Form of Typical Wave Front

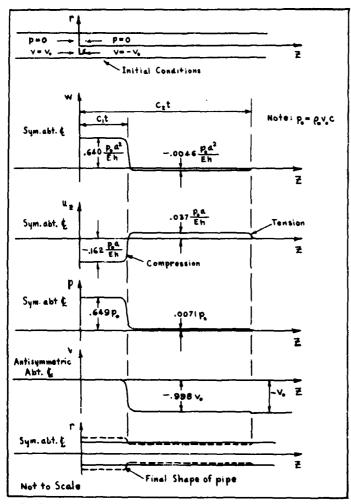


Fig. 6. Values of  $w_1u_{11}p$  and v for a steel tube filled with water,  $\frac{a}{h}$  =62.8. Case of sudden stoppage of flow.

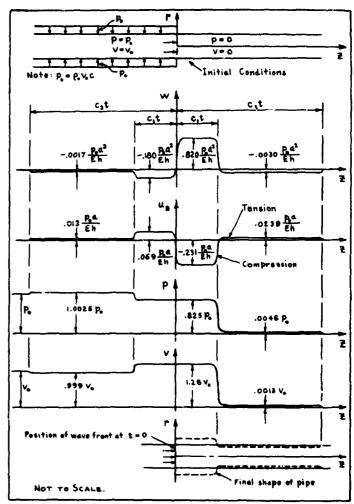


Fig. 5. Values of  $w_1u_2$ , p and v for a steel tube filled with  $wa^{+}e^{-}$ ,  $\frac{a}{h}=62.8$ . Case of initiation of flow.

### DISTRIBUTION LIST

### CONTRACT 266 (08) PROJECT NR360-002

PART A: ADMINISTRATIVE AND LIAISON ACTIVITIES	İ	PART B: THE U. S. DEPARTMENT OF DEFENSE	<u> </u>
ACTIVITIES	•	Office of the Secretary of Defense	
Chief of Naval Research		Research and Development Division	
Department of the Navy		The Pentagon	
Washington 25, D. C.		Washington 25, D. C.	
Attn: Code 418	(1) (2)	Attn: Technical Library	(1)
Code 438	(2)		
		Chief	
Commanding Officer		Armed Forces Special Weapons Project	
Office of Naval Research		The Pentagon	
Branch Office		Washington 25, D. C.	
John Crerar Library Building		Attn: Technical Information Div.	(2)
86 E. Randolph Street	<b>(3.)</b>	Weapons Effects Division	(1)
Chicago Il, Illinois	(1)	Special Field Projects	(1) (1) (2)
Commondan v Officer		Blast and Shock Branch	(2)
Commanding Officer		Common Maria Common 9	
Office of Naval Research		Commanding General	
Branch Office		Field Command	
346 Broadway New York 13, N. Y.	(1)	AFSWP, P.C. Box 5100	(2)
New fork 1), n. 1.	(1)	Albuquerque, New Mexico	(3)
Commanding Officer		ARMY	
Office of Naval Research			
Branch Office		Office of the Secretary of the Army	
1030 E. Green Street		The Pentagon	
Pasadena, California	(1)	Washington 25, D. C.	
•		Attn: Army Library	(1)
Commanding Officer			
Office of Naval Research		Chief of Staff	
Navy #100, % Fleet Post Office	(0)	Department of the Army	
New York, N. Y.	(2)	Washington 25, D. C.	
•		Attn: Development Branch (R&D Div.)	(1)
Director		Research Branch (R&D Div)	(1)
Naval Research Laboratory		Special Weapons Branch (R&D)	<i>a</i> _ s
Washington 25, D. C.	161	Div.)	(1)
Attn: Tech. Info. Officer	(6)		
Code 6200	(1)	Office of the Chief of Engineers	
Code 6205	(1)	Asst. Chief for Military Construction	
Code 6250 Code 6260	(1) (1)	Department of the Army	
	(1)	Eldg. T-7, Gravelly Point	
Code 2029	(1)	Washington 25, D. C.	(- )
Armed Services Technical Information	<b>0</b> 10	Attn: Library Branch	(1) (1)
Agency	OL	Structural Branch (Engr. Div.)	(1)
Document Service Center		Protective Construction Br.	(2.)
Knott Building		(Pl., Engr, & Contracts)	(1)
Dayton 2, Ohio	(5)	Chief Bogogneh and Domoloument	
3-y 001	(5)	Chief, Research and Development	
Office of Technical Services		Department of the Army Washington 25, D. C.	
Department of Commerce		Attn: Special Weapons & Air Defense	
Washington 25, D. C.	(1)	Division	(1)
	• •		14/
		Chief of Engineers	
		Department of the Army	
		Washington 25, D. C.	
		Attn: ENGNB	(2)

Commanding General Aberdeen Proving Ground	Commandant, Marine Corps Headquarters, U.S. Marine Corps	
Maryland	Washington 25, D. C. (1	l)
Attn: Director, Ballistic		
Research Laboratories (	2) Chief, Bureau of Ships	
	Department of the Navy	
Commanding Officer, Engineer	Washington 25, D. C.	
Research & Development Laboratory	Attn: Code 327	2) 1)
Fort Belvoir, Virginia	Code 348	T)
Attn: Chief, Tech Intelligence Br. (	1) Code 371	1)
•	Code 420	2)
Director, Waterways Experiment Station	n Code 423	1) 2) 2) 2) 1)
P. O. Box 631	Code 1442	2)
Vicksburg, Miss.	•	1)
Attn: Library (	1)	
•	Chief, Bureau of Aeronautics	
Commanding Officer	Department of the Navy	
Engineer Research Development	Washington 25, D. C.	
Laboratory		ı)
Fort Belvoir, Virginia (	1) AD-22	ı)
• •	GM (S	1)
Office of the Chief of Ordnance	RS-7 (3	1)
Department of the Army	TD-42 (3	1)
Washington 25, D. C.		
Attn: Research and Materials	Chief, Bureau of Ordnance	
	1) Department of the Navy	
ORDTX - AR (	1) Washington 25, D. C.	•
	Attn: Ad3	1)
Office of the Chief Signal Officer	Re	1)
Department of the Army	<b>Re3</b>	1)
Washington 25, D. C.	Re5	1) 1) 1)
Attn: Engineering & Technical Div. (	1) ReN (	1)
Commanding Officer	Director of Naval Intelligence	
Watertown Arsenal	Navy Department	
Watertown, Massachusetts	Washington 25, D. C.	
		1)
		•
Commanding Officer	Chief, Bureau of Yards and Docks	
Frankford Arsenal	Department of the Navy	
Bridesburg Station	Washington 25, D. C.	·- \
Philadelphia 37, Pennsylvania	Attn: Code D=213 (	1) 1) 1) (1)
Attn: Laboratory Division (	(1) Code D-222 (	1,
0001 0001 0001	Code D-110C (	1)
Office of Ordnance Research	Code D-LLO (	1)
2127 Myrtle Drive	Code D=500 (	,1)
Duke Station	0	
Durham, North Carolina	Commanding Officer and Director	
Attn: Division of Engineering	David Taylor Model Basin	
Sciences	(1) Washington 7, D. C.	,_
314170	Attn: Code 700 (	1
NAVY	Code 720 (	1) (2) (1)
Chief of Namal Oceantians	Code 730 (	,1)
Chief of Naval Operations		
Department of the Navy	Library (	(2)
Washington 25, D. C. Attn: Op 37	(1)	
	(1)	
Op 03EG	(1)	

Commander J.S. Naval Ordnance Laboratory White Oak, Maryland Attn: Technical Library Technical Evaluation Department EE	(2) (1) (1)	Commander Naval Ordnance Test Station Underwater Ordnance Division 3202 E. Foothill Boulevard Pasadena 8, California Attn: Structures Division	(1)
EH R	(1) (1)	Commanding Officer and Director Naval Engineering Experiment Station Annapolis, Maryland	(1)
Director			
Materials Laboratory		Superintendent	
New York Naval Shipyard Brooklyn 1, New York	(1)	Naval Post Graduate School Monterey, California	(1)
•			
Commanding Officer and Director		Commandant Marine Corps Schools	
U.S. Naval Electronics Laboratory San Diego 52, California	(1)	Quantico, Virginia	
Attn: Code 4223	(1) (1)	Attn: Director, Marine Corps	
tun: ode 422)	(-)	Development Center	(1)
Officer-in-Charge Naval Civil Engineering Research and Evaluation Laboratory		AIR FORCE	
U.S. Naval Construction Battalion		Commanding General	
Center		U.S. Air Force	
Port Hueneme, California	(2)	Washington 25, D. C.	
Attn: Code 753	(2) (1)	Attn: Research and Development	4- 5
Director		Division	(1)
Naval Air Experimental Station		Commander	
Naval Air Material Center		Air Material Command	
Naval Base		Wright-Patterson Air Force Base	
Philadelphia 12, Pennsylvania	/n \	Dayton, Ohio	/a \
Attn: Materials Laboratory	(1)	Attn: MCREX-B	(1)
Structures Laboratory	(1)	Structures Division WCOSI	(1) (1)
Officer-in-Charge		***************************************	\-,
Underwater Explosion Research		Commander	
Division		U.S. Air Force Institute of Technology	
Norfolk Naval Shipyard		Wright-Patterson Air Force Base	
Portsmouth, Virginia Attn: Dr. A. H. Keil	(2)	Dayton, Ohio Attn: Chief, Applied Mechanics Group	(1)
	(-/		\-,
Commander		Director of Intelligence	
U.S. Naval Proving Grounds	<b>/</b> 3.\	Headquarters, U.S. Air Force	
Dahlgren, Virginia	(1)	Washington 25, D. C. Attn: P.V. Branch (Air Targets Div.)	(1)
Superintendent		AFOIN-1B2	(1) (2)
Naval Cun Factory			<b>,</b> -,
Washington 25, D. C.	(1)	Commander Air Research and Development Command	
Commander		P. O. Box 1395	
Naval Ordnance Test Station		Baltimore 3, Maryland	<b>.</b> •
Inyokern, China Lake, California	(2.)	Attn: RDMPE	(1)
Attn: Physics Division	(1)		
Mechanics Branch	(1)		

PART C: OTHER GOVERNMENT ACTIVITIES		Director, Operations Research Office	
TI C Atomic Transport		Johns Hopkins University	
U.S. Atomic Energy Commission Washington 25, D.C.		7100 Connecticut Avenue	
	(2)	Chevy Chase, Maryland	
TARREST OF WESSELLON	(2)	Washington 15, D. C.	(1)
Director		Discrete AGETTA Decrete Greater	
National Bureau of Standards		Director, ASTIA Document Service Center	
Washington 25, D. C.		U B Building	
Attn: Division of Mechanics	(1)	Dayton 2, Ohio Attn: DCS-SA (No restricted data to	
	(1)	this addressee)	(1)
Aircraft Structures	(1)	with adm essee)	(1)
Commandant		Dr. Alvin C. Graves, Director	
U.S. Cost Guard		J-Division, Los Alamos Scientific	
1300 E Street, N.W.		Laboratory	
Washington 25, D. C.		P. 0. Box 1663	
Attn: Chief, Testing and		Los Alamos, New Mexico	(1)
	(1)		
-	,	U.S. Atomic Energy Commission	
U.S. Maritime Administration		Classified Technical Library	
General Administration Office Ridg.		Technical Information Service	
Washington 25, D. C.		1901 Constitution Avenue, N.W. Washington, D. C.	
Attn: Chief, Division of Preliminary		Atin: Mrs. Jean M. O'Leary	
Design	(1)	For: Dr. Paul C. Fine	(1)
Notional Administration Res			(-/
National Advisory Committee for Aeronautics		U.S. Atomic Energy Commission	
1512 H Street, N. W.		Classified Technical Library	
Washington 25, D. C.		Technical Information Service	
	(2)	1901 Constitution Avenue, NW	
	(-)	Washington, D. C.	
Director		Attn: Mrs. Jean M. O'Leary	(1)
Langley Aeronautical Laboratory		0. 11 2	
Langley Field, Virginia		Sandia Corporation, Sandia Base	
Attn: Structures Division (	(2)	Albuquerque, New Mexico	4- >
		Attn: Dr. Walter A. MacNair	(1)
Director		U.S. Atomic Energy Commission	
Forest Products Laboratory	<i>-</i> - •	1901 Constitution Avenue, NW	
Madison, Wisconsin	(1)	Washington, D. C.	
Charles Assessment Land Land Land Land		Attn: Dr. John von Neumann	(1)
Civil Aeronautics Administration Department of Commerce		1012 100	(4)
Washington 25, D. C.		Massachusetts Institute of Technology	
Attn: Chief, Airframe and Equipment		Cambridge 39, Massachusetts	
	(1)	Attn: Dr. Charles H. Norris	(1)
•	<b>(</b> -)	7.75 P	<b>\</b> _,
National Sciences Foundation		PART D: INVESTIGATORS ACTIVELY ENGAGED	
1520 H Street, N. W.		IN RELATED RESEARCH	
Washington, D. C.		Professor Temp S. Deedle	
Attn: Engineering Sciences Div. (	(1)	Professor Lynn S. Reedle Fritz Engineering Laboratory	
Waldama 7 Ann Janes a 6 G		Lehigh University	
National Academy of Science		Both ohen Down - 1	/a \
2101 Constitution Avenue			(1)
Washington 25, D. C. Attn: Technical Director, Committee		Professor R. L. Bisplinghoff	
()	- 1	Dept. of Aeronautical Engineering	
Executive Secretary, Committee	. – ,	Massachusetts Institute of Technology	
	1)	Combonidae 20 Magazata Att	(1)
	-/		<b>\-</b> /

Professor H. H. Bleich		Professor P. G. Hodge	
Department of Civil Engineering		Department of Aeronautical Engineering	•
Columbia University		and Applied Mechanics	
- 1 · 1 · 1 · 1 · 1 · 1 · 1 · 1 · 1 · 1	(1)		
New York 27, N. Y.	(1)	Polytechnic Institute of Brooklyn	
		99 Livingston Street	
Professor B. A. Boley		Brooklyn 2, N. Y.	
Department of Civil Engineering			
Columbia University		Professor N. J. Hoff, Head	
New York, N. Y.	(1)	Department of Aeronautical Engineering	
	<b>\-</b> /	and Applied Mechanics	
Preference C F Commiss			
Professor G. F. Carrier		Polytechnic Institute of Brooklyn	
Pierce Hall		99 Livingston Street	/- \
Harvard University	<b>/ &gt;</b>	Brooklyn 2, N. Y.	(1)
Cambridge 38, Massachusetts	(1)		
		Professor W. H. Hoppmann, II	
Professor Herbert Deresiewicz		Department of Mechanical Engineering	
Department of Civil Engineering		Johns Hopkins University	
Columbia University		Baltimore, Maryland	(1)
632 W. 125th Street		per original transf	(4)
	(1)	Desdoner Desso A. Johnston	
New York 27, N. Y.	(1)	Professor Bruce G. Johnston	
		University of Michigan	
Professor Lloyd Donnell		Ann Arbor, Michigan	(1)
Department of Mechanics			
Illinois Institute of Technology		Professor J. Kempner	
Technology Center		Department of Aeronautical Engineering	
Chicago 16, Illinois	(1)	and Applied Mechanics	
	<b>\-</b> /	Polytechnic Institute of Brooklyn	
Professor D. C. Drucker, Chairman			
		99 Livingston Street	/- 1
Division of Engineering		Brooklyn 2, New York	(1)
Brown University			
	/= \		
Providence 12, Rhode Island	(1)	Professor H. L. Langhaar	
·	(1)	Professor H. L. Langhaar Department of Theoretical and	
Providence 12, Rhode Island Professor A. C. Eringen	(1)	Department of Theoretical and	
Professor A. C. Eringen		Department of Theoretical and Applied Mechanics	
Professor A. C. Eringen Department of Aeronautical Engineer		Department of Theoretical and Applied Mechanics University of Illinois	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University	ing	Department of Theoretical and Applied Mechanics	(1)
Professor A. C. Eringen Department of Aeronautical Engineer		Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana	ing	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana Professor W. Flugge	ring (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin	ring (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota	
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station	
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University	ring (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota	
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota Professor E. H. Lee	
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota Professor E. H. Lee Division of Applied Mathematics	
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota Professor E. H. Lee Division of Applied Mathematics	
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois  Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota  Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island  Professor George H. Lee	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis II, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island Professor George H. Lee Director of Research	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis II, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island Professor George H. Lee Director of Research	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois  Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota  Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island  Professor George H. Lee Director of Research Rensselaer Polytechnic Institute	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis II, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island Professor George H. Lee Director of Research	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering	(1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois  Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota  Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island  Professor George H. Lee Director of Research Rensselaer Polytechnic Institute Troy, New York	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University Stanford, California	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois  Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota  Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island  Professor George H. Lee Director of Rasearch Rensselaer Polytechnic Institute Troy, New York  Mr. M. M. Lemcoe	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University Stanford, California  Professor L. E. Goodman	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island Professor George H. Lee Director of Research Rensselaer Polytechnic Institute Troy, New York Mr. M. M. Lemcoe Southwest Research Institute	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University Stanford, California  Professor L. E. Goodman Engineering Experiment Station	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis II, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island Professor George H. Lee Director of Research Rensselaer Polytechnic Institute Troy, New York  Mr. M. Lemcoe Southwest Research Institute 8500 Culebra Road	(1) (1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University Stanford, California  Professor L. E. Goodman Engineering Experiment Station University of Minnesots	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island Professor George H. Lee Director of Research Rensselaer Polytechnic Institute Troy, New York Mr. M. M. Lemcoe Southwest Research Institute	(1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University Stanford, California  Professor L. E. Goodman Engineering Experiment Station	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis II, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island Professor George H. Lee Director of Research Rensselaer Polytechnic Institute Troy, New York  Mr. M. M. Lemcoe Southwest Research Institute 8500 Culebra Road San Antonio 6, Texas	(1) (1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University Stanford, California  Professor L. E. Goodman Engineering Experiment Station University of Minnesota Minneapolis, Minnesota	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois  Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis II, Minnesota  Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island  Professor George H. Lee Director of Research Rensselaer Polytechnic Institute Troy, New York  Mr. M. M. Lemcoe Southwest Research Institute 8500 Culebra Road San Antonio 6, Texas  Professor Paul Lieber	(1) (1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University Stanford, California  Professor L. E. Goodman Engineering Experiment Station University of Minnesota Minneapolis, Minnesota Professor M. Hetenyi	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis II, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island Professor George H. Lee Director of Research Rensselaer Polytechnic Institute Troy, New York  Mr. M. M. Lemcoe Southwest Research Institute 8500 Culebra Road San Antonio 6, Texas	(1) (1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 1019 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University Stanford, California  Professor L. E. Goodman Engineering Experiment Station University of Minnesota Minneapolis, Minnesota Professor M. Hetenyi The Technological Institute	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis 14, Minnesota Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island Professor George H. Lee Director of Rasearch Rensselaer Polytechnic Institute Troy, New York Mr. M. M. Lemcoe Southwest Research Institute 8500 Culebra Road San Antonio 6, Texas Professor Paul Lieber Geology Department	(1) (1)
Professor A. C. Eringen Department of Aeronautical Engineer Purdue University Lafayette, Indiana  Professor W. Flugge Department of Mechanical Engineerin Stanford University Stanford, California  Mr. Martin Goland Midwest Research Institute 4049 Pennsylvania Avenue Kansas City 2, Missouri  Professor J. N. Goodier Dept. of Mechanical Engineering Stanford University Stanford, California  Professor L. E. Goodman Engineering Experiment Station University of Minnesota Minneapolis, Minnesota Professor M. Hetenyi	(1) (1) (1)	Department of Theoretical and Applied Mechanics University of Illinois Urbana, Illinois  Professor B. J. Lazan, Director Engineering Experiment Station University of Minnesota Minneapolis II, Minnesota  Professor E. H. Lee Division of Applied Mathematics Brown University Providence 12, Rhode Island  Professor George H. Lee Director of Research Rensselaer Polytechnic Institute Troy, New York  Mr. M. M. Lemcoe Southwest Research Institute 8500 Culebra Road San Antonio 6, Texas  Professor Paul Lieber	(1) (1)

Professor Hsu Lo School of Engineering Purdue University Lafayette, Indiana	(l)	Professor J. Stallmeyer Department of Civil Engineering University of Illinois Urtana, Illinois	(1)
Professor R. D. Mindlin Department of Civil Engineering Columbia University 632 W. 125th Street New York 27, N. Y.	(1)	Professor Eli Sternberg Department of Mechanics Illinois Institute of Technology Technology Center Chicago 16, Illinois	(1)
Dr. A. Nadai 136 Cherry Valley Road Pittsburgh 21, Pennsylvania	(1)	Professor S. P. Timoshenko School of Engineering Stanford University Stanford, California	(1)
Professor Paul M. Naghdi Department of Engineering Mechanics University of Michigan Ann Arbor, Michigan	(1)	Professor A. S. Velestos Department of Civil Engineering University of Illinois Urbana, Illinois	(1)
Professor William A. Nash Department of Engineering Mechanics University of Florida Gainesville, Florida	(1)	Professor Enrico Volterra Department of Mechanics Rensselser Polytechnic Institute	(2)
Professor N. M. Newmark Department of Civil Engineering University of Illinois Urbana, Illinois	(1)	Troy, New York  Professor C. T. Wang Department of Aeronautical Engineering New York University	(1)
Professor Aris Phillips Department of Civil Engineering 15 Prospect Street		University Heights New York 53, N. Y.  Professor Dana Young	(1)
Yale University New Haven, Connecticut	(1)	Yale University New Haven, Connecticut	(1)
Professor W. Prager, Chairman		Project Staff	(10)
Physical Sciences Council Brown University Providence 12, Rhode Island	(1)	For your future distribution	(10)
Professor E. Reissner Department of Mathematics Massachusetts Institute of Technolo Cambridge 39, Massachusetts	957 (1)		
Professor M. A. Sadowsky Department of Mechanica Rensselaer Polytechnic Institute Troy, New York	(1)		
Professor C. B. Smith College of Arts and Sciences Department of Mathematics Walker Hall University of Florida Gainesville, Florida	,		